Effect of Surface Roughness of Al5052/CFRP Composites on Adhesion and Mechanical Properties†

M.S. LEE¹, S.J. KIM¹, O.D. LIM² and C.G. KANG^{1,*}

¹School of Mechanical Engineering, Pusan National University, Pusan 609-735, Republic of Korea

²R & D Center, Autogen Co., Ltd., 180, Okgucheondong-ro, Siheung-si, Gyeonggi-do, 429-849, Republic of Korea

*Corresponding author: Fax: +82 515181456; Tel: +82 515101455; E-mail: cgkang@pusan.ac.kr

AJC-15812

In this study, Al5052/CFRP composites were fabricated for an automobile component by compression molding process inside a U-channel mold. Al5052 sheet was treated by sand blasting with two different particle sizes. Accordingly, surface roughness (R_a) values of 4.25 and 1.85 μm were obtained for the treated Al5052 sheets. The effect of surface roughness of Al5052 sheets on the adhesion and mechanical properties of Al5052/CFRP composites have been evaluated. Shear lap test and 3-point bending test were conducted. Results showed that the shear load for the composite fabricated by using the treated Al5052 sheets with R_a value of 1.85 and 4.25 μm were three and five times higher than R_a value of 0.73 μm of the composite fabricated by using the untreated sheet. The bending stress of 200 MPa was obtained for the composite fabricated with untreated Al5052 sheets. The bending stress increased to 400MPa when the composite fabricated from treated sheets. However, the bending stress was not influenced by treating a condition through sand blasting.

Keywords: Hybrid composite materials, U-Channel molding, Surface roughness, Porosity, Carbon fiber reinforced plastic.

INTRODUCTION

A recent research trend in the automobile industry is the development of robust, light weight and eco-friendly raw materials that can ensure enhanced safety and fuel efficiency in vehicles, as well as help to reduce emissions from vehicles. Some of these studies are focused on the development of advanced high-strength steel (AHSS)-based hot-press-formed components for automobile parts^{1,2}.

Meanwhile, studies are also being carried out on electric and fuel-cell vehicles. However, such vehicles tend to be heavier than traditional vehicles due to additional components such as electric motors, supplementary batteries and fuel cells installed in them. There are weight-related limits in technological advancements with the currently available high-tensile steel materials. The development of futuristic vehicles requires the use of considerably light materials with higher specific strength and stiffness than currently used steel materials.

The demand for light weight materials in the automobile industry has given rise to the development and application of composite materials such as carbon fiber reinforced plastic (CFRP), besides high-strength aluminum alloy and magnesium.

Carbon fiber reinforced plastic, particularly, has been attracting great interest owing to its superior characteristics, including strength and stiffness^{3,4}. However, CFRP composites have limitations for automotive applications due to their drawbacks of low elongation and fracture toughness, as well as weak bonding between different materials. To overcome such limitations of single-component materials, a study on hybrid composites is currently underway that aims at facilitating welding by mixing CFRP with metallic materials⁵.

The mechanical properties of hybrid composites largely depend on the extent of adhesion achievable between the component materials. Insufficient bonding between different materials lowers crash energy absorption. Furthermore, it is essential to subject CFRP-aluminum (CFRP-Al) junction to a chemical treatment such as etching treatment using phosphoric acid and anodizing prior to bonding⁶, because galvanic effect is likely to accelerate corrosion of aluminum, resulting in the detachment of the junction between CFRP and aluminum. Therefore, extensive research has been carried out to improve the bonding strength between different materials. Al-Zubaidy *et al.*⁷ evaluated the bonding strength of steel-CFRP composites. Chung *et al.*⁸ studied the mechanical properties of

†Presented at 2014 Global Conference on Polymer and Composite Materials (PCM2014) held on 27-29 May 2014, Ningbo, P.R. China

aluminum foam and CFRP. Furthermore, Rhee⁹ treated the surface of T-peel with plasma, enhancing its bonding strength six-fold and shear strength by 33 %. However, aluminum surface treatments, such as chemical etching and plasma treatment, involve high costs, making the mass production of automobile parts using such treatment methods economically infeasible. Therefore, it is crucial to develop a cost-effective surface treatment technology. Furthermore, studies to investigate mechanical properties and adhesion behaviours based on R_a value and epoxy infiltration rate are required. R_a value is the average surface roughness obtained through physical surface treatment. There is also a lack of research on the single-component epoxy infiltration method for the production of morphologically heterogeneous hybrid composites and thickness homogeneity.

The objective of this study was to find the optimal temperature and pressure conditions for the development of CFRP. Bending test was performed to evaluate the mechanical properties of U-channel hybrid composite materials. R_a values were measured to evaluate the sandblast surface treatment of aluminum. The effects of surface roughness on the epoxy infiltration rate and thickness were investigated by measuring the weight per cent of epoxy resin and thinning rate.

EXPERIMENTAL

Production condition for CFRP: The CFRP sheet used in this study was 0.25 mm thick with epoxy comprising 35–40 % of the total weight. The properties of the epoxy are listed in Table-1. Before production of Al5052/CFRP/Al5052 sandwich composites, to examine effect of process parameters on CFRP, samples with five sheets of CFRP were stacked and heated on a hot plate for 0.5 h at different pressures and temperatures. Fig. 1 shows the tensile strength of CFRP according to curing temperature and pressure. Fig. 1(a) and (b) show 140 °C and 0.4 MPa were chosen for the test specimens. The tensile strength was greatest above 0.4 MPa at 140 °C. Fig. 1(b) shows the high tensile strength at 0.4 and 1.2 MPa at 140 °C. However, during specimen production, although the tensile test was favorable for specimens produced under 1.2 MPa, epoxy spilled out and spread to the die. Consequently, the specimen thickness decreased and while the tensile strength was high, the loss of epoxy resulted in a decrease in interfacial strength between CFRP laminate. Therefore, the specimen was produced at 0.4 MPa as the suitable pressure.

TABLE-1
RESIN PROPERTIES

Item	YD-115J	Test method
EEW (g/eq)	175-194	KD-AS-001
Viscosity (cps at 25 °C)	150-500	KD-AS-005
Hy-Cl (wt. %)	0.6-0.9	KD-AS-010
Specific gravity	1.14	KD-AS-040

Fig. 2 shows the microstructure of CFRP specimens produced under different pressures. These microstructures show why the tensile strength of specimens produced under 0.04 MPa was lower than 0.4 and 1.2 MPa. When $P_p = 0.04$ MPa, several pores of 30–40 μm occurred as shown in Fig. 2(a). When $P_p = 0.4$ and 1.2 MPa, the number of pores decreased

(a) Tensile strength according to curing temperature

(b) Tensile strength according to pressure

Fig. 1. Curve of tensile stress and displacement according to temperature and pressure

Fig. 2. Microstructure of CFRP according to pressure

and their size decreased to 10 μm . Zhu *et al.*¹¹ reported that when pressure was higher than 0.4 MPa, porosity became less than 1 % and did not affect mechanical properties.

Shear lap test: Al5052 materials have excellent welding properties and are thus widely used for the production of automobile components. A 0.5 mm thick Al5052 sheet was used in the shear lap test. The mechanical properties of the sheet are listed in Table-2.

TABLE-2
PROPERTIES OF AL5052

Tensile strength (MPa)	Elongation (%)	Brinell hardness (H_b)	Elastic modulus (MPa)
230	12-16	60	70

Fig. 3 shows the cross-section of the bonded interface between the Al5052 sheet and the CFRP with and without sand blasting. To prevent galvanic corrosion of aluminum due to the bonding of different materials, its surface was pretreated

Fig. 3. Cross-section of Al5052 sheet with anodic oxidation coating

with sand blasting, followed by the application of a thin ($< 10 \mu\text{m}$), flexible anodic oxidation coating.

Fig. 4 is the schematic diagram of the experiment. As shown in Fig. 4(a), both surfaces of the aluminum sheet were pretreated with sand blasting and anodic oxidation coating was applied to them, for enhancing the strength of bonding between different materials. To analyze the effect of R_a values on shear strength, specimens were fabricated by bonding a 10 mm CFRP with a 100 mm long and 25 mm wide Al5052 sheet. As shown in Fig. 4(b) and (c), specimens for the shear lap test were fabricated by performing compression molding for 0.5 h under 0.4 MPa in a flat-type die heated to 140°C using a cartridge heater.

Fig. 4. Schematic diagram of experiment

The shear lap test was performed under the MTS conditions of 20 mm/min, by attaching the aluminum flat sheet to the grip part in a flat position. The test was repeated thrice and the middle value of the three tests, *i.e.*, excluding the highest and lowest values, was taken.

U-channel compression molding: U-channel compression molding experiments were performed to investigate the mechanical properties of Al5052/CFRP/Al5052 composite and

the R_a -dependent effects of the epoxy flow. Specimens of the hybrid composite were prepared by employing the following steps. As illustrated in Fig. 4(d), specimens of Al5052/CFRP/Al5052 composite were fabricated by sandwiching five 0.3-mm thick CFRP sheets with two 0.5-mm thick sandblast-surface-treated Al5052 sheets. This way, specimens with total thickness of 2.5 mm ($0.3 \text{ mm} \times 5 + 0.5 \text{ mm} \times 2$) were formed. The width and length of the Al5052 sheet was 130 and 180 mm, respectively.

As shown in Fig. 4(e), U-channel compression molding of the Al5052/CFRP/Al5052 composite was performed for 0.5 h under pressure of 0.4 MPa in a die heated to 140°C with a cartridge heater. This study is a groundwork aiming at developing a technology enabling a mass production process for automobile components. Zhu *et al.*¹¹ reported that voids disappeared almost completely at pressures of 0.4 MPa. In this study, specimens were fabricated through high-pressure compression molding and their physical properties were tested to evaluate if they can be used for the mass production of automobile components.

While flat positions 1, 3 and 5 can be squeezed by a punch in the compression molding process, punch force cannot be applied to vertical positions 2 and 4. Considering this, the punch-die clearance was maintained at 0.7 by setting the tolerance at 1.85 mm. The specimen was thicker (2.5 mm) than the tolerance of the die (1.85). Therefore, a forceful pressure was applied to the specimen during compression molding.

Once the fabrication of the specimens was completed, bending specimens were cut out at positions 1 through 5, as illustrated in Fig. 4(f), to test the R_a -dependent mechanical properties and interfacial adhesion. The bending specimens were of the size $75 \text{ mm} \times 15 \text{ mm}$, matching the span length and compliant with ASTM (D 790-03) standard¹⁰. As shown in Fig. 4(g), a three-point bending test was performed at $V_f = 10 \text{ mm/min}$. Flexural stress was then calculated using eqn. 1,

$$\sigma_f = \frac{3PL}{bt^2} \quad (1)$$

where P , L , b and t denote flexural load, span length, width and thickness, respectively.

As shown in Fig. 4(h), after evaluating the mechanical properties of five pairs of symmetrically cut square specimens of size $10 \text{ mm} \times 10 \text{ mm}$, at positions 1 through 5, their R_a -dependent epoxy flows and weight per cent values were measured. As shown in Fig. 4(i), the epoxy amount-dependent epoxy flow was evaluated by comparing the pre-molding and post-molding epoxy weight percents using a scale. The epoxy weight per cent of each square specimen was then calculated using eqn. 2,

$$W_r = \frac{\rho_r V_r}{\rho_r V_r + \rho_f (1 - V_f)} \times 100 \quad (2)$$

where W_r , ρ_r , ρ_f , V_r and V_f denote epoxy weight per cent, epoxy density, CFRP density, epoxy volume and CFRP volume, respectively.

The change in thickness of each specimen, which depended on their corresponding epoxy weight per cent, was calculated using eqn. 3.

$$\gamma = \frac{t_o - t}{t_o} \times 100 \quad (3)$$

where γ , t_o and t denote thinning rate, pre-mold thickness and post-mold thickness, respectively.

RESULTS AND DISCUSSION

Surface treatment with sand blasting: Unlike ionic plasma etching, sand blasting is a sputtering-based physical etching method, wherein atoms or molecules are detached from the surface of the material-Al5052, in case of this study-by transferring kinetic energy from an abrasive particle aspirated by air pressure. This kind of physical etching yields the same results as ionic etching.

The R_a value of the surface of Al5052 varies depending on the size of the abrasive particle. The increase in the R_a value is proportional to the increase in the size of the abrasive particle and *vice-versa*, due to the detachment of same-sized aluminum surface particle.

Fig. 5 shows the images of the surface profiles of sand blast-treated or untreated aluminum specimens. The images were captured using an atomic force microscope (AFM). Table-3 outlines the conditions for the AFM observations.

Fig. 5. 3-Dimensional interactive display according to surface roughness with sand blasting

TABLE-3 EXPERIMENTAL CONDITION FOR AFM TEST AND AFM EXPERIMENTAL MACHINE			
Magnification	Measurement mode	Sampling (nm)	Array size (μm)
20	VSI	496.74	640 × 480

As shown in Fig. 5(a), the R_a value of an untreated specimen was found to be 0.73 μm and the surface profile of the specimen

hardly showed any marked differences, except for a few spots. Fig. 5(b) and (c) show the AFM images of the sandblast-treated specimens with R_a values of 1.85 and 4.25 μm, respectively. From the results of the sand blast treatment, it can be concluded that the R_a value is directly proportional to the deviation in the surface roughness of aluminum. R_a values of 0.73, 1.85 and 4.25 μm yielded R_q values of 0.93, 2.61 and 5.43 μm, respectively. R_q is the square roots of mean peak-to-valley distances.

Fig. 6(a) through (c) represent surface roughness through a two-dimensional chart, in which the x-axis denotes the surface roughness. As shown in Fig. 6(a), the untreated specimen with R_a value of 0.73 μm yielded the R_p (highest peak) value of 1.40 μm and the R_v (lowest valley) value of -1.86 μm. Fig. 6(b) and (c), representing the sandblast-treated surface profiles, show that the increase in the R_a value from 1.85 to 4.25 μm yielded the increase in the R_p value from 3.71 to 6.76 μm and the decrease in the R_v value from -5.25 μm to -8.59 μm. The epoxy infiltration volume increases with the increase in the sizes of the detached aluminum particles, due to the increase in the R_t value. The R_t value increased from 1.40-8.97 μm and then to 15.35 μm, with the increase in the R_a value from 0.73 to 1.85 and then to 4.25 μm, respectively.

Fig. 6. Roughness of 2-dimensional line section with sand blasting

Shear lap test according to surface roughness: Fig. 7 illustrates displacement graphs of shear load and surface roughness. Shear load was applied until an Al-CFRP specimen, with a 10 mm lap joint, failed. While the untreated specimen with R_a value of 0.73 μm showed clean disjointed surfaces, the sandblast-treated aluminum surfaces were left with considerable amounts of epoxy due to greater surface roughness. It was found that with the increase in the roughness of the surface, the volume of infiltrated epoxy left on the aluminum contact surface after failure also increased. The maximum shear load increased from 900 to 3,000 N and then to 5,000 N, while the displacement rose from 0.5 to 0.8 mm and then to 1.27 mm. Meanwhile, the R_a value jumped from 0.73 μm (untreated) to 1.85 μm and then to 4.24 μm (surface-treated), demonstrating

Fig. 7. Shear load and displacement according to surface roughness

three- and five-fold increases in maximum shear loads, respectively.

Flexural stress according to surface roughness: Fig. 8(a) through (c) illustrate the relationship between the position-dependent flexural stress and the displacement of the U-channel specimens, depending on surface roughness. As shown in Fig. 8(a), the flexural stress of the untreated specimen was found to be about 500 MPa at flat positions 1, 3 and 5 and as low as 150–200 MPa at vertical positions 2 and 4. In contrast, the surface-treated specimens with R_a value of 1.85 and 4.25 μm demonstrated a considerable increase, in flexural stress to 400 MPa, at positions 2 and 4, as shown in Fig. 8(b) and (c). While it was observed that the sideward flexural stress of the surface-treated specimens was twice that of the untreated specimen, the flexural stress at flat positions 1, 3 and 5 did not reach the level of the untreated specimens. The weaker adhesion of the untreated specimen at positions 2 and 4 with respect to positions 1, 3 and 5 was due to the lack of application of pressure at the vertical positions, unlike at the flat positions. Unlike shear load, no R_a -dependent differences were observed in the flexural stress of the two surface-treated specimens at positions 2 and 4.

Fig. 9 illustrates the R_a -dependent mechanisms of the delamination of the specimens under shear and flexural loads. The R_a value and CFRP epoxy infiltration of the surface-treated Al5052 specimens are positively related. As shown in Fig. 9(a) and (b), the higher the R_a value, the greater is the resistance by the infiltrated epoxy against delamination, when the specimen is exposed to shear load. On the other hand, when flexural load is applied vertically, no deep-infiltrated epoxy is observed, as shown in Fig. 9(c) and (d). This explains the insignificant R_a -dependent differences in the flexural loads at the vertical positions, as shown in Fig. 8(b) and (c), unlike shear load, which increased proportionally with the increase in the R_a value, as shown in Fig. 7.

Fig. 10 shows the side views, by position and R_a value, of the bending specimens after the bending test. At all positions, 1 through 5, one bending side of the untreated specimen was detached after the bending test. Furthermore, while both surface-treated specimens resisted the bending test at positions 1, 3 and 5, they got detached at positions 2 and 4. This concludes that the bonding of the interfaces at vertical positions 2 and 4 is weaker than the adhesion at flat positions 1, 3 and 5, even after the sand blast-surface treatment.

(a) Untreated $R_a = 0.73$

(b) Treated $R_a = 1.85$

(c) Treated $R_a = 4.25$

Fig. 8. Correlation flexural stress and displacement according to surface roughness

Fig. 9. Mechanism according to surface roughness in shear lap test and flexural test

*S.R.: Surface roughness, P.N.: Position number

Fig. 10. Side view of bending specimens after bending test according to surface roughness

Fig. 11 shows the microstructure of the bending specimens, by position, after the bending test. The compression molding was expected to leave few voids¹¹ because the process was performed under a pressure of 0.4 MPa. However, voids of size 30–40 μm occurred at positions 2 and 4 due to the lack of continuously applied pressure, while there was no evidence of any void at positions 1, 3 and 5. The voids generated at vertical positions 2 and 4 were assumed to be the cause of deteriorated mechanical properties.

Fig. 11. Microstructure of bending specimens after bending test

Epoxy weight per cent and thinning rate: The initial epoxy weight per cent of prepreg is 35 % and that of the Al5052-sandwiched specimen is 16.7 %. Fig. 12 shows the R_a -dependent weight per cent at each position. The epoxy weight percents at vertical positions 2 and 4 of the untreated specimen were found to be 5 and 7 %, respectively. A large amount of epoxy leaked out of the die during the molding process due to the clearance of 0.7 between the punch and the lower die. The epoxy weight percents at flat positions 1 and 5, after the molding process, were found to be 11 and 11.5 %, respectively. This was due to the compression-induced leakage of the epoxy from the die. Position 3 showed the highest epoxy weight per cent of 15 % as the leaked epoxy was compensated by the epoxy inflow induced by gravity. This resulted in a decrease of 1.7 % in the epoxy weight per cent at position 3 from the initial epoxy weight per cent.

Fig. 12. Epoxy weight percent according to surface roughness

Surface roughness and epoxy weight per cent were found to be positively related at all positions, 1 through 5. At position 3, the epoxy weight per cent of the untreated specimen with R_a value of 0.73 μm and the sandblast-treated specimen with R_a value of 4.25 μm were found to be 15 and 18.7 %, respectively, both higher than the initial epoxy weight per cent of 16.7 %. This implies that as the roughness of the aluminum surface increases, the area of the aluminum-epoxy interface increases and the epoxy infiltrates deeper across the aluminum surface. At all positions, the epoxy weight per cent increased with the increase in the R_a value, demonstrating that an increase in surface roughness increases the epoxy weight per cent.

Fig. 13 shows the R_a -dependent thinning rates. The thinning rates at vertical positions 2 and 4 of the untreated specimen with R_a value of 0.73 μm are 18.2 and 17.5 %, respectively. High thinning rates at the vertical positions were results of high amount of epoxy leakages at them. Position 3, which was found to exhibit the highest epoxy weight per cent, showed the lowest thinning rate.

Fig. 13. Thinning rate according to surface roughness

It was found that surface roughness was positively related with epoxy weight per cent, but negatively related with thinning rate. Particularly, at position 3, while the thinning rate was 2.3 % at the R_a value of 0.73 μm , it was -10.2 % at the R_a value of 4.25 μm , i.e., the thinning rate decreased with the increase in the R_a value. The decrease in the thinning rate was due to the increase in the epoxy weight per cent. As shown in Fig. 8, the post-molding epoxy weight per cent was found to be more than the initial or pre-molding epoxy weight per cent. Given

that epoxy flow influences thickness, it is necessary to evaluate and control epoxy flow behaviour to maintain uniform thickness.

Consideration of defect volume through X-ray computed tomography: Fig. 14 shows X-ray computed tomography (CT) of specimens after the bending test according to surface roughness. X-ray CT data was obtained after the Al5052 was removed. Porosity was 0.5-1.2 % irrespective of surface roughness. The porosity data show that hot compression molding method was highly effective for fabricating Al5052/CFRP/Al5052 composites without autoclaving using a vacuum process. Zhu *et al.*¹¹ reported that CFRP composites presented good mechanical properties when the porosity was less than 1 %. Defect volumes were generally 0.1-0.8 mm³. Some defect volumes of 0.2-0.5 mm³ were observed in the untreated specimen as shown in Fig. 14(a). The number and volume of defects increased with increasing surface roughness. Particularly, when surface roughness was 1.2 μm as shown in Fig. 14(c), the number and volume of defects was larger than in the other specimens and many defects of 0.4-0.6 mm³ were distributed. This phenomenon might have occurred because the high surface roughness of the Al5052 impinged on the CFRP surface and epoxy was removed along with the Al5052. The porosity of the CFRP surface increased for this reason. However, porosity of 1.2 % was satisfactory and surface defects were only slightly higher than in other specimens. It is possible to surmise that the internal defects of the 3 specimens [Fig. 14 (a-c)] were nearly the same.

Fig. 14. X-Ray computed tomography (CT) of specimens after bending test according to surface roughness

Fig. 15 shows that X-ray CT data of specimens after bending test according to position number. When position numbers were 1, 3, 5, porosities were 1.2, 1.1 and 0.9 %, respectively. However, porosity of side position was 2.4-3.2 % (Position number 2, 4) as shown in microstructure (Fig. 11), because enough pressure was not applied to these positions. These porosities were higher than 1 % and high porosity was reason of low mechanical properties as shown in Fig. 8.

Fig. 16 shows numerical data of defect volume obtained through X-ray CT of the specimens. The numerical data of specimens with position number 1, 3 and 5 had a similar trend. However, number of the volume of 0.05-7 mm³ defects were

Fig. 15. X-Ray CT data of specimens after bending test according to position number

Fig. 16. Numerical data of defect volume obtained through X-ray CT of the specimens

more than the defect volume of other positions at side positions (position number 2 and 4), generally. The numerical data from the X-ray CT results accorded closely with X-ray CT of CFRP according to position numbers as shown in Fig. 15.

Conclusion

- With the increase in surface roughness, aluminum-epoxy interface increases, resulting in considerable increase in shear load. The sand blast-surface-treated specimens with R_a values of 1.85 and 4.25 μm yielded three- and five-fold shear loads, respectively.

- The flexural stress of the untreated specimen in vertical positions 2 and 4 was as low as 150-200 MPa, while that of the surface-treated specimens increased to 400 MPa.

- The higher the R_a value, the deeper is the epoxy infiltration and the higher is the post-molding volume of epoxy. The increase in the volume of epoxy infiltration resulted in the increase in thickness. Thus, to maintain uniform thickness, it is necessary to control epoxy flow.

- As surface roughness was higher, the number and volume of defects was larger and many defects volumes were distributed on surface of CFRP. Mechanical properties were satisfactory at position number 1, 3 and 5, when porosity was around 1 % but when porosity was higher than 1 %, mechanical properties were affected at side position.

ACKNOWLEDGEMENTS

This study was supported by the Advanced Technology Center Project (10039031, The Development of the Manufacturing Process for the High Strength Steel and Composite Materials) funded by the Ministry of Knowledge Economy (MKE, Korea) and National Research Foundation of Korea (NRF) grant funded by the Korea Government (MSIP) through GCRC-SOP (No. 2011-0030013).

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